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Application No. 09/862,979

Page 2, in between lines 6 and 8, insert:

BRIEF DESCRIPTION OF THE DRAWINGS

Page 2, in between lines 21 and 23, insert:

DETAILED DESCRIPTION OF PREFERRED EMBODIMENTS

Page 4, lines 11-20, delete current paragraph and insert therefor:

In this configuration, radar signals are either diffused by the shroud 10 or absorbed by special coatings applied to it. At the same time, exhaust gas 44 issuing from the exit of the gas turbine engine 6 passes down the duct means 2,4 and through the nozzle 8. Secondary air 46, taken from the airflow outside the shroud is directed through the shroud annulus. This secondary air 46 is at a lower temperature and pressure than the exhaust gas 44 and so cools the duct means, 2,4, nozzle 8 and shroud 10 thus reducing the infra-red signature as well as protecting any sensitive coatings applied to the shroud. The secondary air 46 is then directed by the shroud 10 exit aperture to form a sheath of cool air around the exhaust gas 44 leaving the nozzle. This covering of cool air masks the infra-red signature of the exhaust plume.

IN THE CLAIMS:

Please replace claim 1 as follows:

1. (Amended) A shrouded nozzle arrangement for a gas turbine engine exhaust gas comprising a gas turbine, an exhaust nozzle, a duct means for providing communication of exhaust gas between the gas turbine and the exhaust nozzle, and a shroud which encloses the nozzle and duct means, the shroud having an exit aperture through which, in use, the exhaust nozzle discharges, wherein the exhaust nozzle is translatable from a first position, wherein an exit plane of the nozzle lies upstream of the exit aperture of the shroud, to a

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